

専門科目 (TG010001)

流体力学II

Fluid Mechanics II

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境界層 1 (Boundary Layer)

■ 境界層の例：平板間の流れ（ジェット、ポアズイユ）

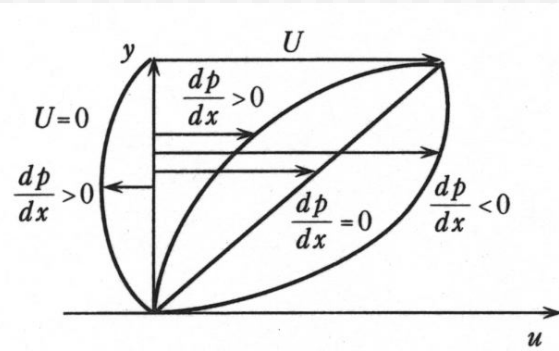


図 5.7 速度分布

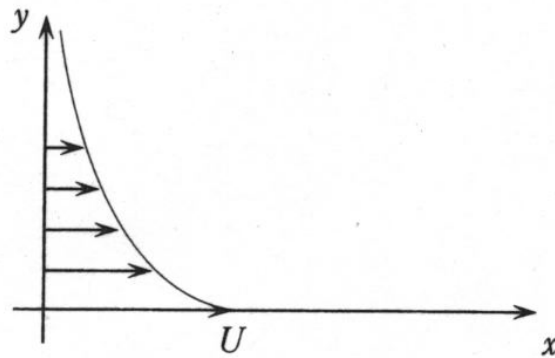
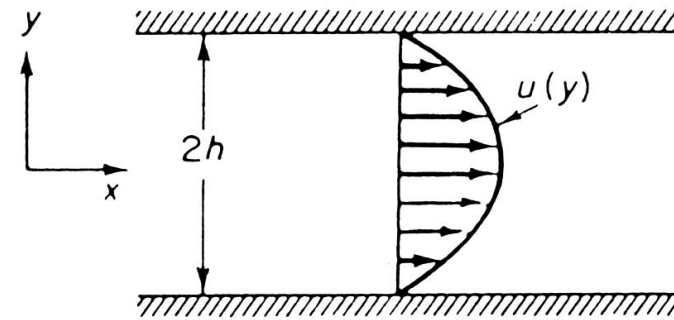


図 5.10 レイリー問題

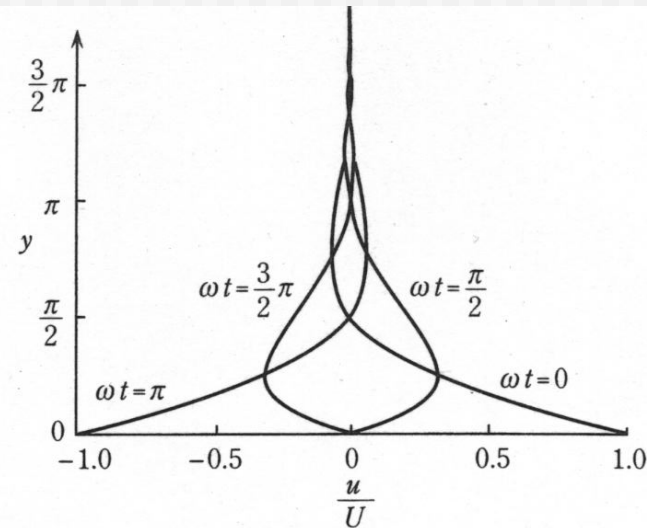
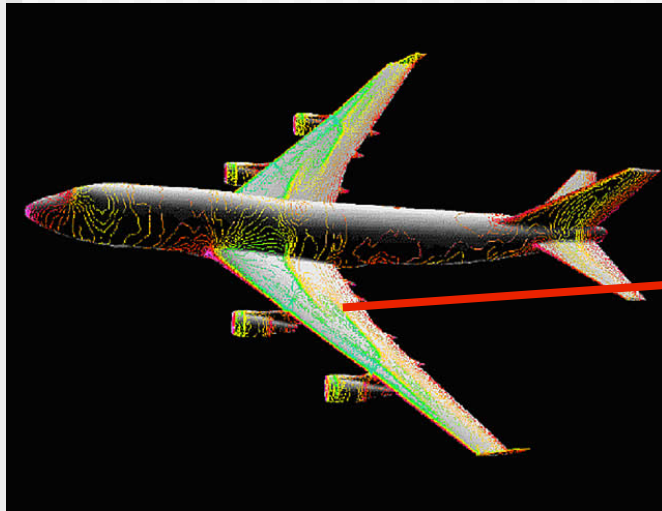


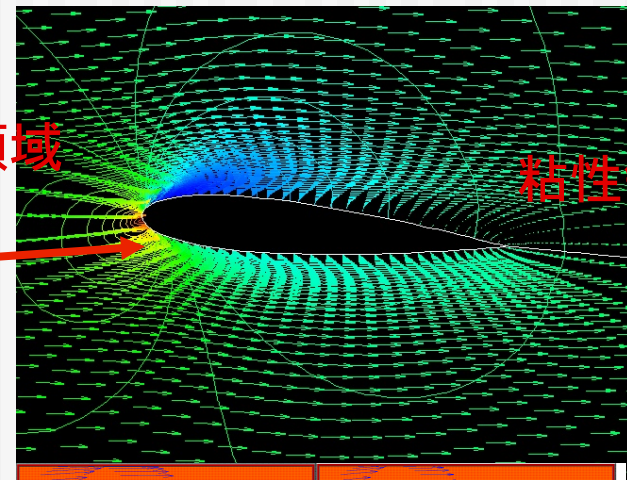
図 5.13 振動平板による流れ

境界層 2 (Boundary Layer)

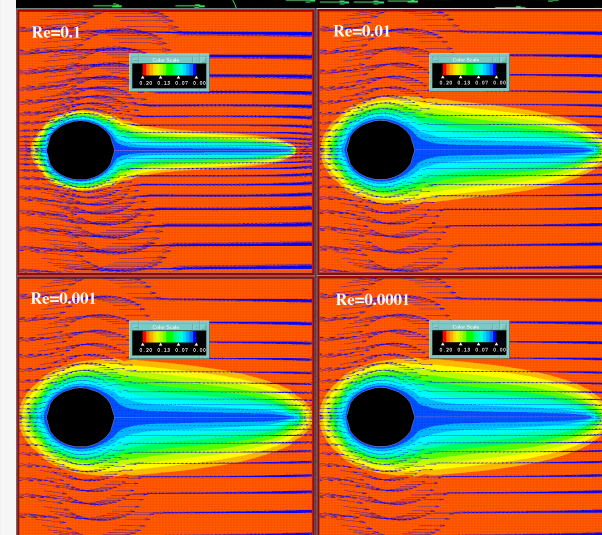
■ 境界層の例：工学問題



非粘性領域

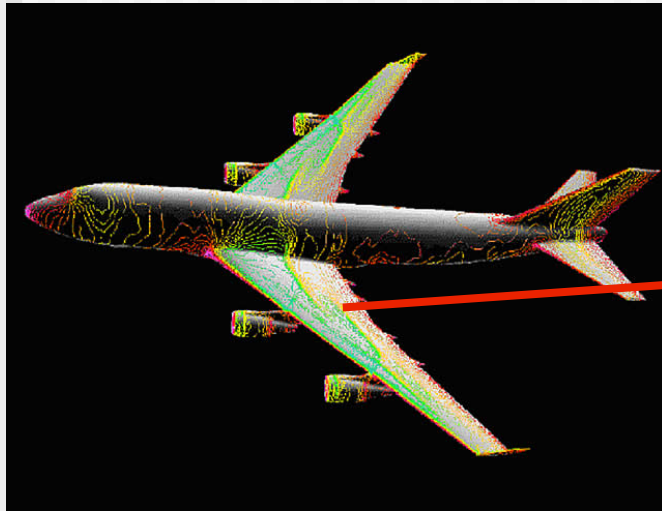


粘性領域

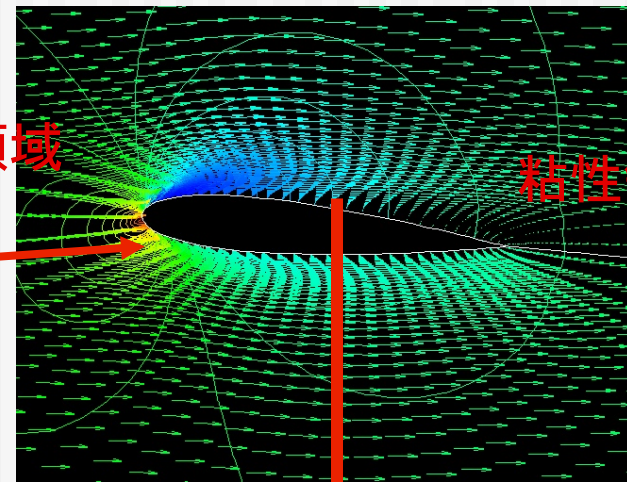


境界層3 (Boundary Layer)

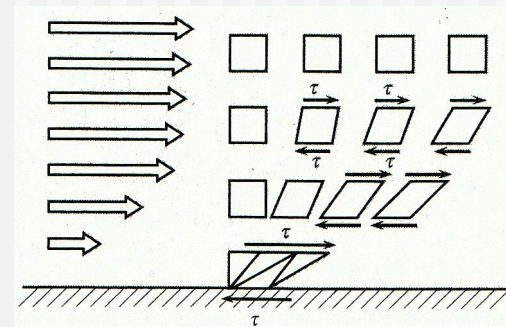
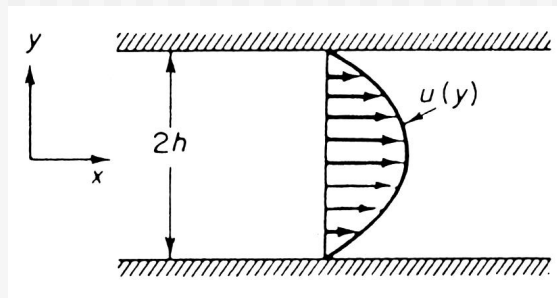
■ 境界層の例：工学問題



非粘性領域



粘性領域

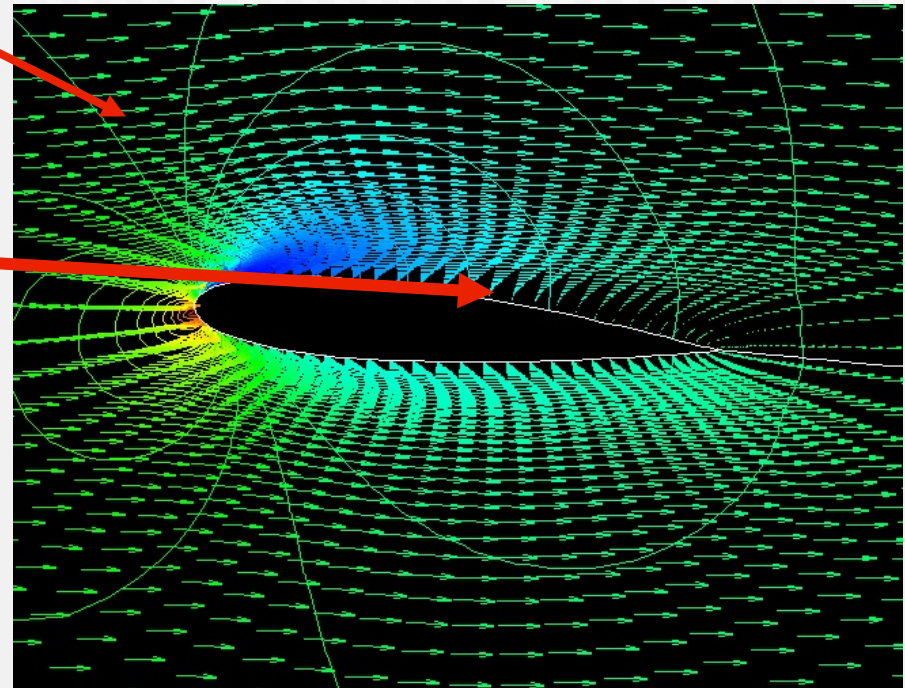


粘性によるせん断応力～変形

境界層方程式 (Boundary Layer Equation)

- 流体支配方程式：

- 境界層近似：
 - @物体から離れた所
 - @物体近傍



流体の支配方程式 (Governing Equations)

- 連続の式 :

$$\frac{\partial \rho}{\partial t} + \rho \operatorname{div} V = 0$$



$$\frac{\partial \rho}{\partial t} + \rho \left(\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} \right) = 0$$

圧縮性流体

$$\frac{\partial \rho}{\partial t} = 0, \quad \frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0$$

非圧縮性流体

- ナビエ・ストークス方程式 :

$$\frac{DV}{Dt} = F - \frac{1}{\rho} \operatorname{grad} p + \nu \nabla^2 V$$

$$\frac{\partial u}{\partial t} + \frac{\partial u}{\partial x} u + \frac{\partial u}{\partial y} v = -\frac{1}{\rho} \frac{\partial p}{\partial x} + \nu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right) + F_x$$

$$\frac{\partial v}{\partial t} + \frac{\partial v}{\partial x} u + \frac{\partial v}{\partial y} v = -\frac{1}{\rho} \frac{\partial p}{\partial y} + \nu \left(\frac{\partial^2 v}{\partial x^2} + \frac{\partial^2 v}{\partial y^2} \right) + F_y$$

プロファイル積分法

(Integration(Karman) Equation of Boundary Layer)

- 境界層内の運動量保存則：

$$BCs: y = 0 \rightarrow u = 0; y = \delta \rightarrow u = U(x)$$

FluxBalance :

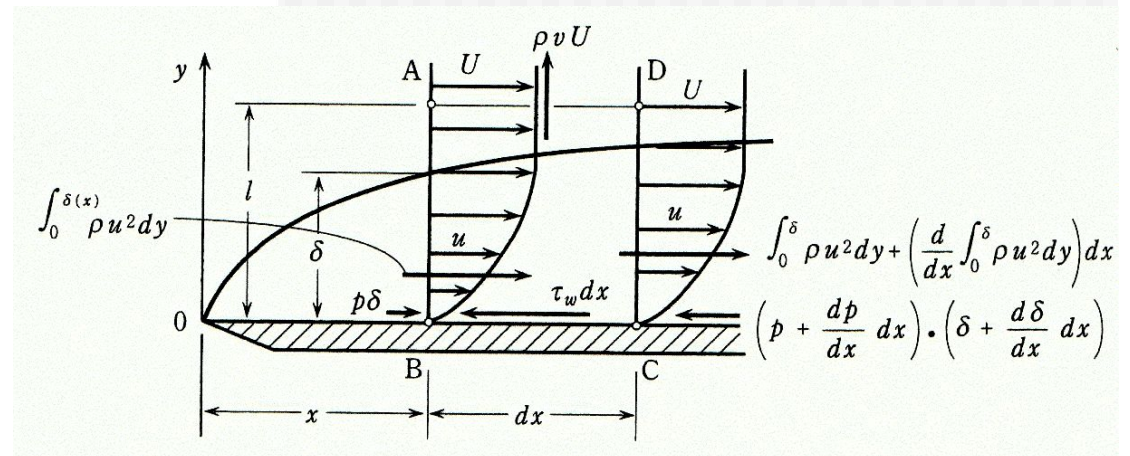
$$AB: \int_0^{\delta(x)} \rho u^2 dy$$

$$CD: \int_0^{\delta(x+\delta x)} \left[\rho u^2 + \frac{d(\rho u^2)}{dx} dx \right] dy$$

$$AD: \rho v U dx$$

$$BC: \tau_w dx = \mu \frac{du}{dy} dx$$

$$\frac{d}{dx} \int_0^{\delta} (U - u) u dy - \frac{dU}{dx} \int_0^{\delta} u dy = \nu \frac{\partial u}{\partial y} - \delta U \frac{dU}{dx}$$



境界層近似 (Boundary layer Approximation)

- 境界層近似：

- @境界層は薄い、 $\delta/L = \text{小さい}$

- $$\delta/l \approx O(0) \rightarrow x \propto l, y \propto \delta$$

- @境界層を横切る方向の速度成分 $v \sim \delta$

- $$v/U \approx O(0) \rightarrow v \propto \delta \Rightarrow u \propto l, v \propto \delta$$

- @圧力は境界層内(y方向)で一定

- $$\frac{\partial p}{\partial y} \approx O(0)$$

- @粘性項のうち、x方向の2回微分は省略できる

- $$\frac{\partial u}{\partial x} / \frac{\partial u}{\partial y} \approx O(0) \rightarrow \frac{\partial^2 u}{\partial x^2} \approx O(0)$$

境界層方程式の導出1

(Formation of Boundary Layer Equation)

- 代表的物理量：

代表長さ=L 代表速度=U 代表時間=T=L/U

- 無次元化：

$$(u,v,w) = U(u^*,v^*,w^*), (x,y,z) = L(x^*,y^*,z^*), t=t^*L/U \quad \text{Re} = \frac{UL}{\nu}$$

$$\frac{\partial u}{\partial t} + \frac{\partial u}{\partial x}u + \frac{\partial u}{\partial y}v = -\frac{1}{\rho} \frac{\partial p}{\partial x} + \nu \left(\frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right)$$

$$\frac{U}{L/U} \frac{\partial u^*}{\partial t^*} + \frac{U^2}{L} \left(\frac{\partial u^*}{\partial x^*} u^* + \frac{\partial u^*}{\partial y^*} v^* \right) = -\frac{1}{\rho L} \frac{\partial p}{\partial x^*} + \nu \frac{U}{L^2} \left(\frac{\partial^2 u^*}{\partial x^{*2}} + \frac{\partial^2 u^*}{\partial y^{*2}} \right)$$

$$\frac{\partial u^*}{\partial t^*} + \frac{\partial u^*}{\partial x^*} u^* + \frac{\partial u^*}{\partial y^*} v^* = -\frac{\partial p^*}{\partial x^*} + \frac{1}{UL/\nu} \left(\frac{\partial^2 u^*}{\partial x^{*2}} + \frac{\partial^2 u^*}{\partial y^{*2}} \right) \leftarrow p^* = \frac{p}{\rho U^2}$$

流力2

$$\uparrow x, u = O(l), y, v = O(\delta), \frac{1}{\text{Re}} = O(\delta^2) \uparrow$$

境界層方程式の導出2

(Formation of Boundary Layer Equation)

- 境界層方程式とその特徴： 有次元化

$$(\mathbf{u}, \mathbf{v}, \mathbf{w}) = U(\mathbf{u}^*, \mathbf{v}^*, \mathbf{w}^*), (\mathbf{x}, \mathbf{y}, \mathbf{z}) = L(\mathbf{x}^*, \mathbf{y}^*, \mathbf{z}^*), t = t^* L/U \quad \text{Re} = \frac{UL}{\nu}$$

$$\frac{\partial u^*}{\partial x^*} u^* + \frac{\partial u^*}{\partial y^*} v^* = -\frac{\partial p^*}{\partial x^*} + \frac{1}{\text{Re}} \left(\frac{\partial^2 u^*}{\partial y^{*2}} \right) \rightarrow \frac{\partial u}{\partial x} u + \frac{\partial u}{\partial y} v = -\frac{1}{\rho} \frac{\partial p}{\partial x} + \nu \frac{\partial^2 u}{\partial y^2}$$

$$p + \rho \frac{U^2}{2} = \text{Const} \Rightarrow \frac{dp}{dx} + \rho U \frac{dU}{dx} = 0$$

$$\frac{\partial u}{\partial x} u + \frac{\partial u}{\partial y} v = U \frac{dU}{dx} + \nu \frac{\partial^2 u}{\partial y^2} \quad x, u = O(l), y, v = O(\delta), \frac{1}{\text{Re}} = O(\delta^2)$$

$$\frac{\partial p}{\partial y} = 0$$

境界層近似の制限

(Limits of Boundary layer Approximation)

- 境界層近似の制限：
@Re数が十分小さい、
粘性境界層が厚い場合

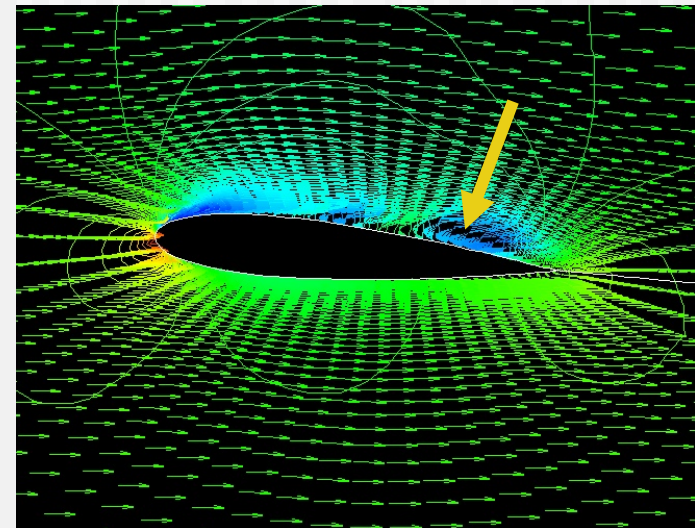
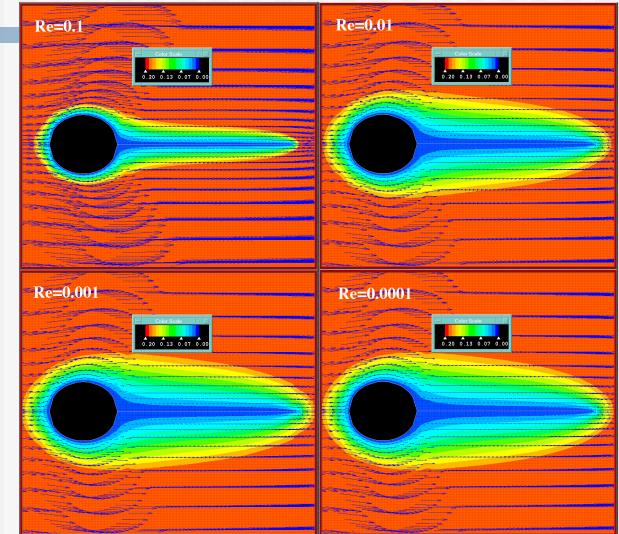
$$\frac{1}{\text{Re}} \neq O(\delta^2)$$

- @平板などの先端付近の流れ
(x方向変化有)

$$\frac{\partial u}{\partial x} \neq O(\delta)$$

- @表面から流線が離れる剥離点近傍より
下流などの領域で法線方向速度が
小さくない場合

$$v \neq \delta$$



ストークス流れ1 (Stokesian Flow)

- 支配方程式：

$$\text{div}V = 0 \quad \frac{\partial V}{\partial t} + V \cdot \nabla V = F - \frac{1}{\rho} \text{grad}p + \nu \nabla^2 V$$

- ストークス近似：移流項を無視
- 一様流れの中に置かれた球まわりの流れ：

$$\text{BodySurface} : u = v = w = 0$$

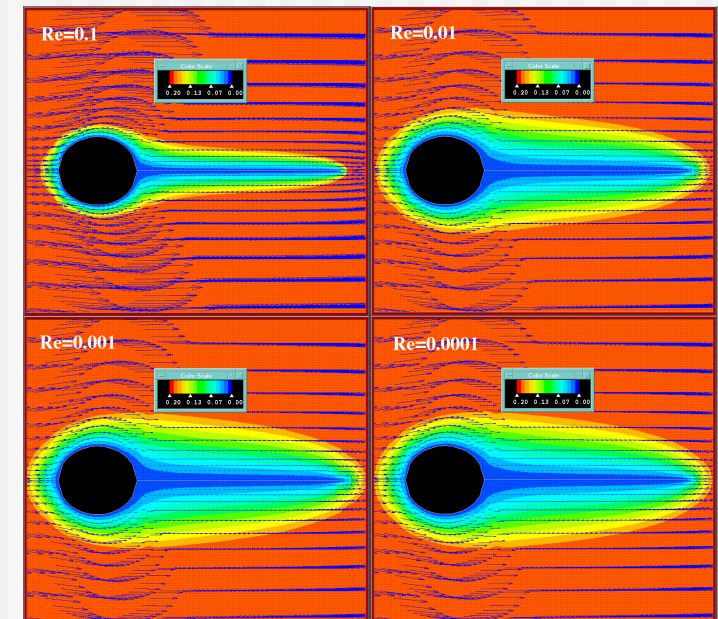
$$\text{OutsideBoundary} : u = U, p = p_\infty$$

$$v_r = U \cos \theta \left(1 - \frac{3/2 \cdot a}{r} + \frac{1/2 \cdot a^3}{r^3} \right)$$

$$v_\theta = -U \sin \theta \left(1 - \frac{3/4 \cdot a}{r} - \frac{1/4 \cdot a^3}{r^3} \right)$$

$$p = p_\infty - \frac{3}{2} \mu \frac{U \cos \theta}{a}, \tau_{r\theta} = \frac{3}{2} \mu \frac{U \sin \theta}{a}$$

流力2



ストークス流れ2 (Stokesian Flow)

- 抵抗の式：

$$\begin{aligned} Drag_x &= -\int_0^\pi \tau_{r\theta} \sin\theta ds - \int_0^\pi p \cos\theta ds \\ &= 4\pi a\mu U + 2\pi a\mu U = 6\pi a\mu U \end{aligned}$$

$$C_d = \frac{Drag}{\frac{1}{2}\rho U^2 \pi a^2} = \frac{24}{Re}$$

- 2次元円柱まわりのストークス流れが存在しない
ストークスのパラドックス

@オゼーン近似：

$$\frac{\partial V}{\partial t} + U \frac{\partial V}{\partial x} = F - \frac{1}{\rho} \text{grad} p + \nu \nabla^2 V$$

流力2

$$\begin{aligned} Drag_x &= -\int_0^\pi \tau_{r\theta} \sin\theta ds - \int_0^\pi p \cos\theta ds \\ C_d &= \frac{Drag}{\frac{1}{2}\rho U^2 \pi a^2} = \frac{24}{Re} \left(1 + \frac{3}{16} Re\right) \end{aligned}$$